

# BHE UAV Family

## Unmanned Aerial Vehicle System

UAV Development at  
BHE Bonn Hungary Electronics Ltd.



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### 1 Applications and design guidelines

The demand on flying is old as humankind. As the mankind had conquered the air and flying has been possible a path to unlimited revolutionary new application areas has been opened. These applications can be sorted in the following groups: transportation of person or goods, warfare, security and surveillance, scientific experiments and data acquisition. Traditionally these applications require manned aircrafts equipped by highly skilled pilots and staff. Therefore the operational and maintenance cost are still high despite of the fact that the cost of aircrafts and aerospace components are decreasing. As the technology and electronic computers advanced the requirement of cutting the cost of many of these applications created a new class of air systems, the unmanned aerial systems. Common characteristics of these systems that the operational costs are lower by a decade, operating does not require highly skilled pilots therefore these UAV systems are excellent choices for cost sensitive public applications which require professional solutions. In the following list some of our target applications can be read:

- » air surveillance and photography
- » disaster monitoring (flood, forest fire or tsunami monitoring)
- » security and defense (border control, area patrolling, monitoring industrial areas, oil and gas pipelines, electrical transmission lines, search for lost people, mass monitoring)
- » archaeology
- » agricultural applications (inland water monitoring, drought damage monitoring, pest infection monitoring)
- » nature protection ( animal movement monitoring, forest monitoring)
- » meteorological and climate research

There are many UAV solutions on the market in different scales and complexities. These can be classified from the viewpoint of effective range, system complexity, covered application area.

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Our company targeted the development of a medium-range UAV system with high system complexity and superior technical and functional capabilities. In the following list we summarize the main capabilities and design guidelines of BHE UAV system:

### » reliability:

This principle is the most important property of our system. We know that there is no second chance in a UAV mission. A system failure is unacceptable at these applications. Not only our reputation is jeopardized but even human life is endangered. Therefore all system components and software are designed for stable operation, long operating life and for harsh environmental conditions. We try to minimize the human factor in the system for reducing the risk of operator errors. The system has several emergency automatisms for the case of for example battery failure or communication jamming. The UAV can autonomously operated but there is anytime the possibility of manual intervention when unforeseen circumstances occur. Therefore all UAV telemetry data (position, speed, height, battery charge, etc..) can be monitored real-time. BHE UAV contains a separate, high dynamic range CMOS pilot camera sensor to provide continuous manual supervision capability on the flight route.

### » security:

It is important that no incompetent, unauthorized party can gain important mission data out of the system. Therefore all the communication and stored data is secured by authorization and encryption.

### » fully automatic flight capability:

The system is capable to autonomously fly on a route of waypoint, placed over a map. The autopilot stabilizes the aircraft and navigates it on the flight route with the aid of GPS sensor. No manual intervention is needed under normal operating conditions.

### » robust communication:

The usability of the system depends on the stable communication between the UAV and the ground monitoring station. In most cases the application requires real-time data transmission from the aircraft at a high data rate, such as real-time video streams. BHE communication system is designed for long range, low latency and security. The communication is flexible and adaptive to always maximize the communication channel capacity. The modulation method is very robust and performs well in line-of-sight and non-line-of-sight situations.

### » own design:

The whole system, every key component and subsystem is designed by us or is fully under our control which fact provides us outstanding flexibility to user demands and requests and makes us independent from international market limitations

### » easy, user friendly operation

All system component is straight-forward, ergonomic, easy to operate and easy to replace or service.

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### » cost effective:

The system requires no infrastructure for operation. The operation requires no special skills, all necessary knowledge is provided during a short training. The maintenance costs are low, the aircrafts, sensors, other components are designed for long operating life.

### » versatile:

The system is flexible, supports different sensor and aircraft types and can be extended in the future. The aircraft is suitable to the most applications and can be equipped with even with user defined payload (with size, weight and power consumption limitation).

### » effective range of 15 km

### » stabilized dual camera sensors:

easily reconfigurable visible, color camera sensor and thermal camera sensor is offered with our system mounted on a gyro-stabilized pan-tilt pedestal. Color camera sensor has zoom feature and provide high resolution, color images. Thermal camera sensor is working in the 8-12 um range and provides digital images with 16 bit thermal resolution.

### » high dynamic range, color, CMOS pilot camera for continuous manual supervision over flight path

### » accurate, reliable MEMS sensors for inertial stabilization, speed and altitude measurement

### » Galileo-ready GPS sensor for accurate navigation

### » support: World-wide support with quick reaction time is provided by us. We are open for unique requests and development as well.

In the following section the description of the system components will be presented.

## 2 UAV system and components

BHE UAV system is a complete real-time surveillance system with all the hardware and software components and support services which are required by the modern customers. The system has an air frame with BXAP13 aircraft which has an action radius more than 20 km with onboard autonomous navigation, high resolution video camera and video transmitter and recorder units. The BXGC10 ground control station with computers, antenna system and launch catapult is for real-time control and monitor the BXAP13 airplanes, to aid the launch and recovery. The ground control station can be installed in the field or in a vehicle.

### 2.1 Air-frame

BHE UAV system incorporates a high capacity, versatile, durable air platform which is carefully designed and developed specifically for unmanned surveillance purposes. It is small but powerful. Thanks to the careful aerodynamic design and composite technology BHE UAV airplane has high endurance (up to one hour flight time), high payload capacity (3 kg of user defined payload) combined with the advantages of electrical propulsion.

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It has a maintenance-free brushless electrical engine for higher reliability, low noise emission and high power. It has a medium speed and altitude capability which makes it usable for wide range of applications. The energy storage element is the latest technology Lithium-Polymer rechargeable battery with low weight, high capacity and long lifetime. Due to the high power propulsion it needs lower take-off distance and can climb in a higher rate. The aerodynamic properties of BHE UAV grant excellent gliding capability and high stability under windy conditions. The hull is made of composite fiber and by default it has a low visibility gray color. If needed other colors are available.



Figure 1. BXAP13 aircraft

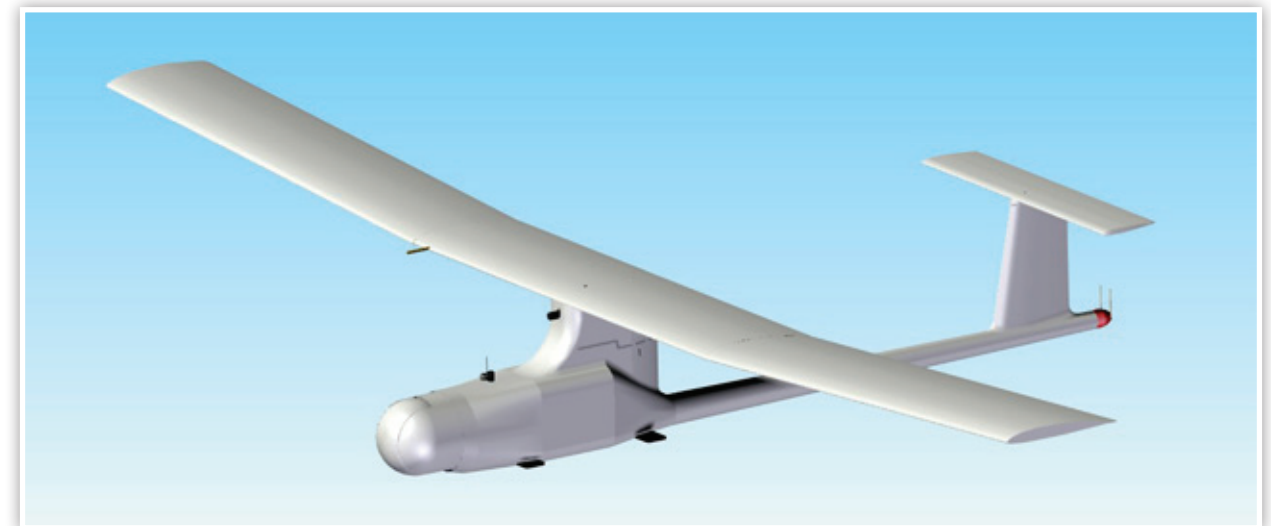


Figure 2. 3D CAD design of BXAP13 aircraft

It is small and easily transportable in a factory provided hard case. All necessary components are preinstalled on the aircraft. After unpacking only 5 major components have to be connected together by a simple "click" or by fixing a screw. Neither special tool, nor complex adjustment procedure are required. The process is so simple that one person can do it in 2 minutes.

BHE UAV aircraft is so lightweight that even a person can toss it to the sky. But because the rotating propeller the risk of injury is high and can be very dangerous therefore we do



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not suggest this kind of start. We provide a catapult which has electrical propulsion. The catapult can be installed on ground or on a vehicle. It can be transported in 1m long pieces and can be assembled by fast connectors. Before take-off the BHE UAV airplane should be placed in the catapult. Before take-off visual inspection of the system must be made by the user by the aid of a pre-flight checklist. The system also executes self-test procedures.

Take-off can be initiated from software on the ground control station. During take-off the BHE UAV climbs with full engine power on a straight path until a safe altitude is reached. After completing the mission the landing procedure must be executed to recover the BHE UAV. We do not use parachutes to recover the airplane, because this method eliminates all control over the aircraft. BHE UAV lands by slipping on its belly on the ground and remains controllable during the whole landing procedure. This type of landing requires minimal ground preparations before landing, because no great pieces of ground objects should block the path of the airplane to protect the payload which is placed in the front of the airplane. At landing the wind direction is critical, because the speed of the aircraft must be kept as low as possible without stalling. Airplane stability can be maintained only if the airplane is landed against wind direction. The wind direction can change rapidly and the wind direction meter at the ground control station can not give precise information for landing. Therefore the landing procedure is half-automatic, which means that the stabilization of the plane and regulation of landing speed is made by the autopilot, but the elevator and rudder has to be controlled manually with the aid of the autopilot. The landing path can be better controlled this way. This kind of landing needs some practice but by the aid of the autopilot it is much easier than the full-manual mode. The landing is aided also by a fixed position pilot camera image and the flight instruments on the navigator software on the ground control station.

### Physical characteristics

**Air Speed:** 50-110 km/h

The speed range is carefully selected for general surveillance applications. The speed is referenced to the air medium and not to the ground. The speed over ground value (SOG) is influenced by wind direction and magnitude. Higher air speed can be reached during dipping, in that case 150 km/h is the maximal air speed.

**Maximal Wind Magnitude:** 40 km/h

To guarantee the aircraft stability in windy conditions it is advised not to fly in higher wind speeds.

**Ceiling:** min. 1000 m, depending on configuration

The ceiling value represents an altitude range referenced to the starting altitude level which grants optimal flying properties to the aircraft. Beyond this value the overall efficiency of the propulsion and the endurance may be degraded. Ceiling value mostly depends on the propeller because the air density decreases with altitude thus propeller blades with

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different higher pitch values needed for higher altitudes. The propeller can be changed by the user and makes it possible to fly at higher altitudes. The propeller mounted to the motor shaft determines the endurance, climbing rate, flight speed and altitude range, however it can be easily replaced by the user we suggest to use BHE certified propellers otherwise we can not guarantee these parameters. BHE certified propellers have high quality, matched to the aircraft and propulsion system and balanced perfectly for low noise output and low vibration.

**Climbing rate:** 2 m/s

The powerful electrical propulsion and the carefully selected propeller gives this capability to the BHE UAV.

**Study of climbing capability:** Starting position is at 200 m ASL.

In this test we studied the climbing properties during manual and autonomous control. It can be seen that the autopilot can perform near as good as our highly skilled pilot.

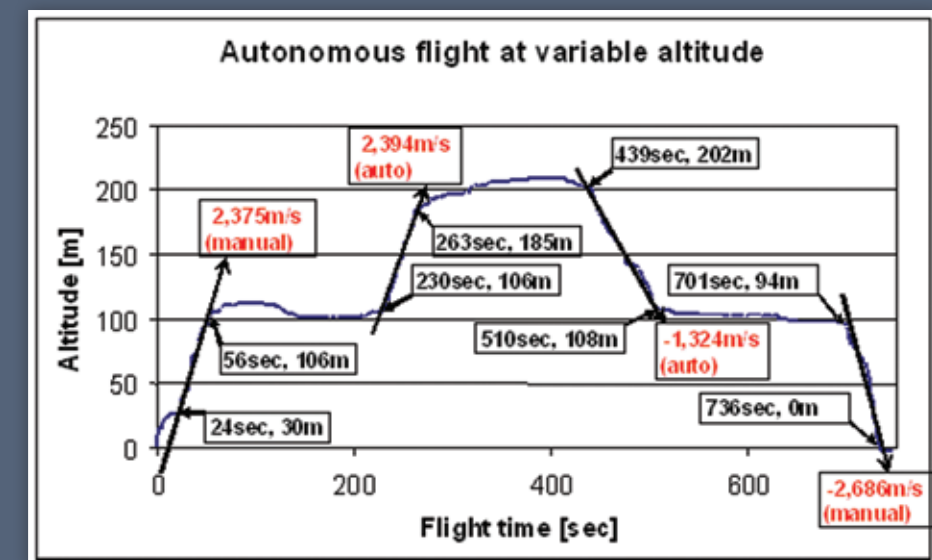


Figure 3. Climbing performance

**Wing Span:** 3.15 m

The wing profile is carefully designed for good aerodynamic properties and provides superior gliding capability to the BHE UAV and grants longer mission time by decreasing battery consumption. The wing is splitted in three pieces for easy transportation and can be easily assembled and mounted on the aircraft by screw.

**Length:** 1.5 m

This dimension provides good stability to the BHE UAV. Because the body of the plane is

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made in one piece this is the largest dimension which determines the size of transportation hard case.

### Payload capacity: 3 kg

User can place any payloads in the front of BHE UAV. Payloads can be attached and detached easily by screws. Stable connection to the aircraft is guaranteed by uniform mechanical interface. The payloads and battery can be replaced in seconds. We provide some standard payloads such as high resolution visible light video camera and high resolution thermal camera.

### Take-off Weight: 7.5 kg

The low weight of the BHE UAV makes it perfect for low speed cruising and surveillance. Even a person could toss it in the air but the risk of injuries is very high in this case that is why we provide catapult for take-off aid.

### Propulsion: electric brushless motor

Maintenance-free, brushless, electric motor with internal cooling system grants reliability, high climbing rate, long endurance and long maintenance-free period for the BHE UAV. Low vibration compared to combustion engines provides low noise output and better video picture quality. The propeller mounted to the motor shaft determines the endurance, climbing rate, flight speed and altitude range, however it can be easily replaced by the user we suggest to use BHE certified propellers otherwise we can not guarantee these parameters. BHE certified propellers have high quality, matched to the aircraft and propulsion system and balanced perfectly for low noise output and low vibration.

### Battery: Lithium-Polymer

Reliable, high capacity, low weight batteries gives the excellent endurance and climbing capabilities to BHE UAV. These batteries can be easily replaced between missions for quick take-off. Although these batteries require less maintenance the health of the battery determines the aircraft performance. Therefore we provide a battery charger and conditioner unit with each BHE UAV system to make it easier for the user to keep the batteries in good working order.

### Endurance: up to 1 hour

The excellent propulsion system and aerodynamics of the BHE UAV make this value possible. Endurance strongly depends on the battery type and health condition furthermore on flight route, wind conditions and mission altitude above the starting position. Often varying altitude during mission, higher wind speed or higher mission altitude require higher engine power and consume battery power decreasing mission time. BHE UAV can provide this endurance time under most conditions but it cannot be guaranteed under any conditions.

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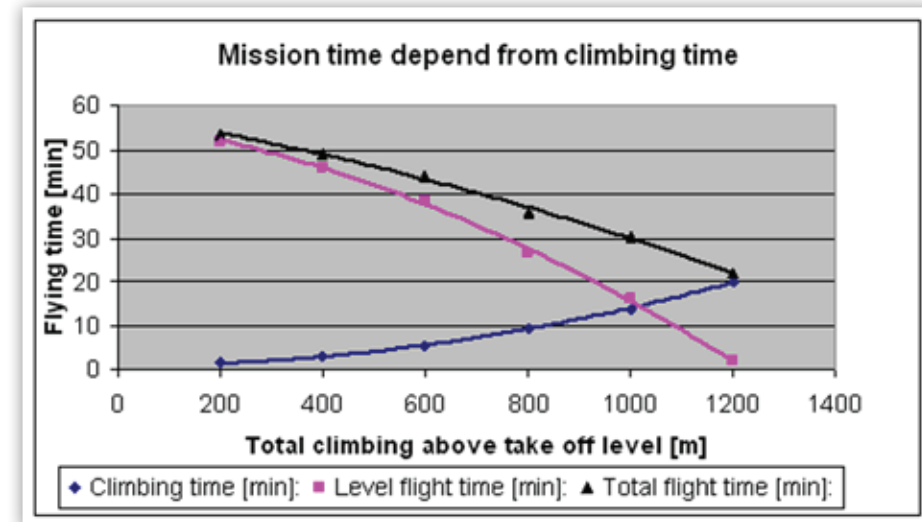


Figure 4. Endurance performance versus altitude

### 2.2 Autopilot

BHE UAV autopilot is a microcontroller based autonomous navigation unit, which uses GPS for navigation and inertial sensor for stabilization. Our sensors are using modern MEMS technology, which yields small size, low weight, long operation time and high precision. Our autopilot, such the other components of BHE UAV system is designed for maximal reliability, precision and easy operability. Reliability is reached by using a dedicated ARM core microcontroller which does no other task only navigation, MEMS sensors which contain no conventional mechanical components and by careful software design which means that the software is tested even under extreme conditions and it can handle unexpected events. The GPS sensor has high sensitivity.

#### Main technical features:

- » 32 bit ARM core processor: we have selected ARM core for our microcontroller, because it is the market leader industrial solution for microcontrollers, it is very stable and reliable
- » Galileo-ready GPS module (4 Hz update rate): a high precision, high sensitivity receiver, which is capable to process the new European Galileo signals
- » 3D inertial navigation sensor: MEMS sensor which is reliable and accurate under most extreme conditions and provides precise orientation information over the aircraft motion
- » MEMS pressure sensors: the accurate measurement of airspeed (by Pitot tube) and barometric altitude is also important for navigation and it is done by high quality, miniature pressure sensors
- » onboard flight data logging: every sensor data, output data, warnings, alarms, unsolicited events, command and telemetry data is logged onboard for security and user specific purposes

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- » 8 PWM servo output channel: these outputs give the control signals for the high precision, high torque servos which are connected to the airplane controls
- » General purpose digital control outputs: user specified digital control lines for special payloads

### Main services:

- » Autonomous onboard navigation: our autopilot is able to stabilize and navigate the BHE UAV airplane on a defined route where the user specifies GPS coordinates and altitude over ground for each waypoints
- » No need for adjustments: every autopilot is adjusted to the BHE UAV in the factory, there is no need for the user to make any adjustments
- » On-board pre-flight self test: before flight the autopilot executes a test procedure and does not allow take-off if major failure is detected in the system
- » Onboard logging: every sensor data, warnings, alarms, output signal, command and telemetry is logged by the autopilot for security and user post-processing purposes
- » Flight route can be defined up to 500 waypoints: the memory of the microcontroller allows the handling of such high number of waypoints, between the waypoints the autopilot navigate the airplane on a straight line, complete flight route can be uploaded or downloaded from the ground control station
- » Real time route/flight parameter modification: the programmed waypoints can be modified during mission
- » Real time manual flight: some of the aircraft controls can be controlled manually, the operator can select which controls he wants to control and the aircraft will respond immediately, this function grants high degree of freedom over the capabilities of the aircraft, however it has the risk of losing the aircraft due to manual failure
- » Preset flight maneuvers: in case of special flight events, such as loss of communication or an important sensor a predefined protocol will be executed by the autopilot to try to drive the aircraft back to the starting point
- » Health monitoring: the autopilot continuously evaluates the system performance, sensor data validity, sensor malfunction, communication link quality, unusual airplane motion, battery capacity, engine failure, etc and sends warning or alarm indications back to ground and stores these also on the onboard flash card

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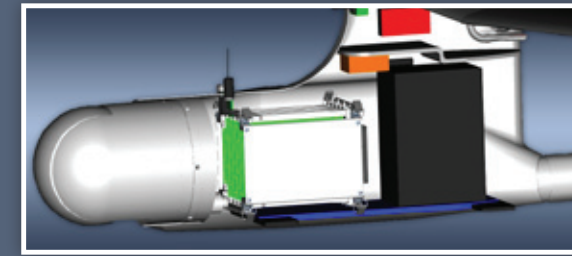


Figure 5. 3D CAD model of onboard electronics and sensors

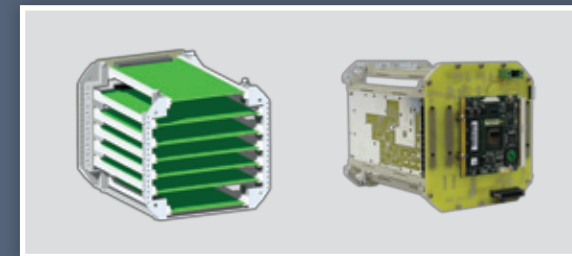


Figure 6. 3D CAD design and photograph of rack for onboard electronics

### 2.3 Onboard data logger

The BHE UAV airborne system contains an on-board data logger which receives video signals and other telemetry from onboard units and telecommand data from ground control, stores the video in high quality on flash disk.

### 2.4 Payloads

BHE provides two standard payloads for wide application areas. These payloads have a common mechanical interface which enables fast payload replacement. The battery is also can be replaced comfortably. These payloads have visible light or thermal cameras placed on an electromechanically stabilized gimbal system. It provides very stable video image in a wide range of situations, during different flight maneuvers. The camera payload has basic mechanical protection against impacts.

#### Visible Light Camera Payload, main technical specifications:

- » High resolution color camera: more than 520 TV line resolution
- » Optical zoom: more than 20x optical zoom is available
- » Autofocus: automatic and manual focusing mode is supported
- » Digital picture stabilization: electronically fine stabilization of the image is made by onboard DSP processing
- » 25 frame per second



### Thermal Camera Payload, main technical specifications:

- » 8-12 um wavelength
- » uncooled microbolometer sensor
- » 384 x 288 pixel resolution
- » 25 frame per second
- » 16 bit dynamic range
- » better than 100 mK thermal resolution



Figure 7. 3D CAD design of visible and dual-sensor camera payload

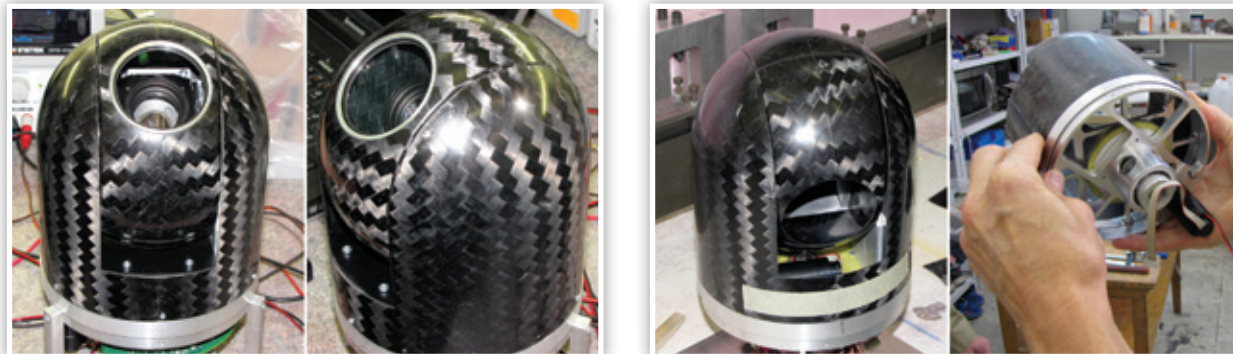


Figure 8. Photograph of the camera head

### User Defined Payload:

Any kind of user defined equipment can be placed on the uniform mechanical interface. BHE supplies empty compartment for fitting any payload by the user. The weight of payload must be less than 3 kg. The proper weight balance of the aircraft must be maintained and the payload must not influence the aerodynamics of the aircraft.

### 2.5 Communication

BHE UAV system has adaptive, long range, reliable, robust wireless radio communication data link. The link operates on the C band (4.5-4.9 GHz). The range is better than 5 kilometers and will be extended further. The data radios using highly interference tolerant spread spectrum modulation. Video is transmitted to ground on a high speed channel, command and telemetry is transmitted via lower speed but more robust channel. Robustness is also enhanced the error detection and correction coding. The radio link is adaptive so the radio parameters are depend on the distance between the aircraft and the ground control station. The communication is protected by additional scrambling for higher security. Communication system is developed jointly with Mobile Innovation Centre of Budapest University of Technology and Economics.

### 2.6 Ground Control Station

The BHE UAV ground control station is a modular system which consist of a computer database server, one or more computer workstations, communication equipment with antennas and mast, GPS and wind sensors and the launch catapult. The database server and the workstation can be the same computer. The system can be mounted on a vehicle or it can be installed on ground. Every component can be disassembled for easier transportation. A transport hard case is provided for the system. The purpose of the ground control station is to provide all hardware elements for controlling the BHE UAV, aid the take-off and store every mission specific data. The real-time control and monitor application runs on a workstation which can be placed far to the ground control station only a network connection must be set up between them. The database server stores every important data for further analysis. The user can replay the missions, search for a specific date and time. The data can be exported from the database and the airborne-recorded data can be imported when the aircraft is recovered. To provide a reliable and stable data link between the BHE UAX aircraft the antennas are placed on a rotating mast. The wind meter measures the wind magnitude and direction to aid the positioning of the catapult against the wind direction for a better take-off. The BHE UAV have to be inserted in the catapult and slipped into the guiding rails. An electrical motor accelerate the aircraft to the take-off speed.



Figure 9. Ground Control Station



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Figure 10. Launch catapult

### 2.7 Software and mission management

The ground control station has one or more computers, which are organized in a local area network. The heart of the system is the database server, which receives, processes and stores every pieces of information about the mission (telemetry data, command data, video streams, etc.). Furthermore the system contains the navigation client software, surveillance client software, server management software and other software utilities, which can run on the server computer or on separate workstations. The workstations are in connection with the server, by database query the actual data flow or a previously stored data flow can be retrieved from the server and it is displayed at the operator's workstation.

After powering on the server and booting is done the database server is operational and it connects to the communication equipment and via the data link discovers the BHE UAV aircrafts within the action radius of the GCS.

To initiate a take-off a navigator software should be started at the server or at a connected workstation. The navigator software requires identification of the operator what it is required for unambiguously determine the access rights. After successful identification the operator has the choice to start a new mission, join as observer to an existing mission, replay a previously recorded one or search for a specific date or location.

If he chooses a previously recorded mission the navigator client software queries the mission list and the operator can select which mission we would like to replay. Searching the archives will display a list of mission entry point with the specified initial date or location. Joining to an existing mission as observer allows the operator to get the route parameters, telemetry and video data, but cannot control the aircraft or affect anyhow the mission.

If the operator chooses to initiate a new mission the navigator client software asks whether the operator would like to plan a new mission route or load a previously saves mission route. By selecting the loading of previously saved mission the operator can review the mission and

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to start a take-off he shall request the control of an aircraft. The navigator client software queries the server for the list of all aircrafts associated with the server. The list contains the operating status of the aircraft (online, on mission, offline, under repair, etc), which aircraft is within communication range, who controls the aircraft now, does the operator allowed to access and control the aircraft. If the control request is successful the operator have control over the aircraft and get all telemetry data. He can upload the mission route to the aircraft and the system verifies that the navigator client software route is identical to the one in the aircrafts memory. If the server indicates that the aircraft self-test is passed the take-off can be initiated. During mission the control over the aircraft is achieved via navigator client software, all the data is archived onboard and at the ground control station server, the camera payload control and visual observation is made with the surveillance client software. The softwares are developed jointly with John von Neumann Faculty of Informatics of Budapest Tech.

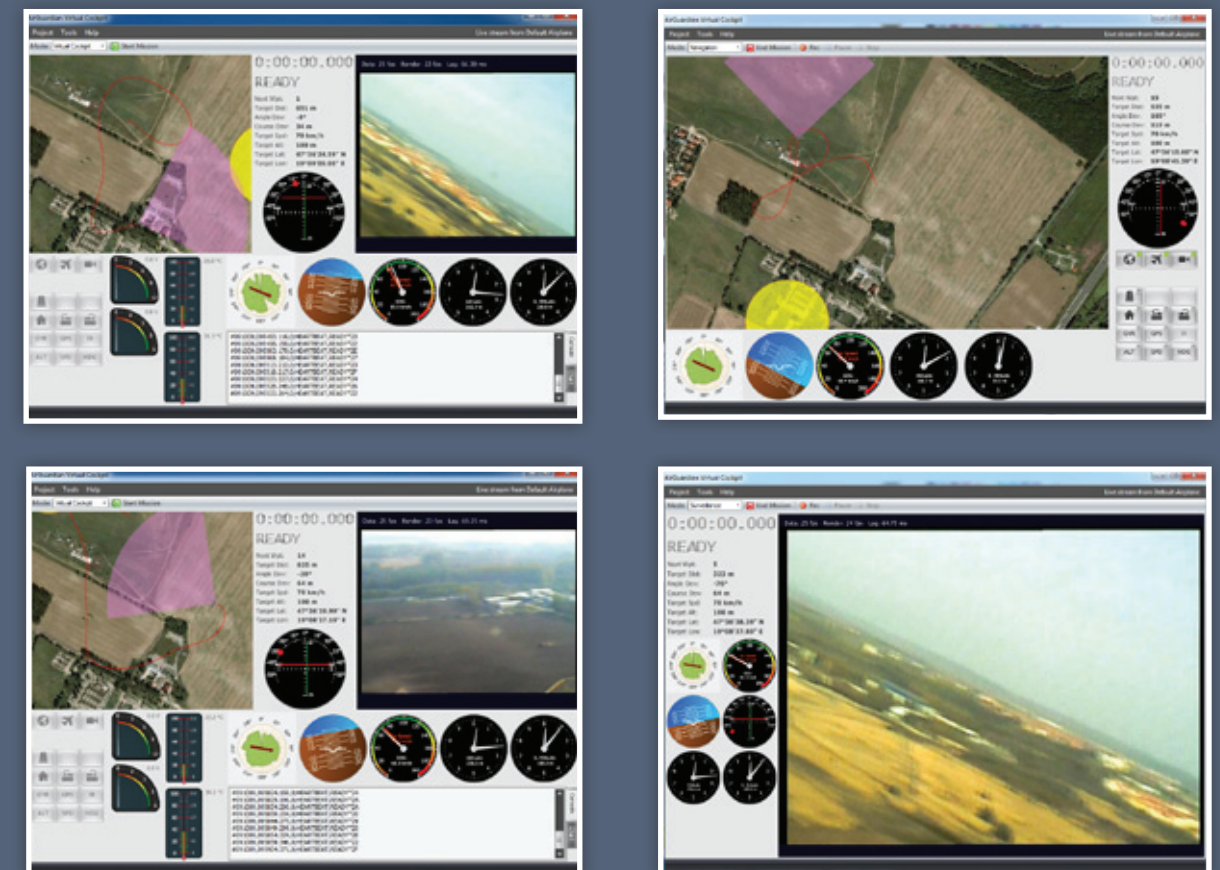


Figure 11. Screen layout of the current control software

### Mission planning

A mission route can be created offline and can be modified online. A mission route contains all information for the BHE UAV to be able to complete the mission fully automatically without need of manual supervision or control. The 3D coordinates of the starting point, the



ending point and the middle waypoints must be defined, point-by-point or with the aid of the advances features of the navigator client software. Along the route the camera orientation can be defined for fully automatic visual observation applications. Special control functions can be associated with the waypoints (activating digital outputs, etc). Flight routes can be saved or loaded from the server.

### Main functions of the navigator client software

#### Mission planning:

- » Mission planning
- » 3D visualization of flight route (with zooming, rotation, panning) with map below the track
- » Handling of raster map or digital map
- » Handling of POIs, special areas (habited area, forbidden area, danger zone, state border, etc)
- » Load, save mission routes

#### Mission supervisioning:

- » Display of map, aircraft identifier, position, camera orientation and FOV, POIs for every available aircrafts (only one can be controlled)
- » Flight instruments (altitude, speed)
- » Aircraft alarm indications

#### Manual control:

- » Joystick input for controlling the aircraft
- » A set of manual controls can be defined (between full-automatic to full-manual)
- » Alarms

### Main functions of surveillance client software:

- » Control of aircraft camera (camera orientation, zoom, focus)
- » Camera information on OSD or separate text window (GPS coordinate, altitude, camera orientation, zoom level (FOV), etc)
- » Get coordinate (calculate the GPS coordinates of a designated point)
- » Hold coordinate (control the camera to hold the center of picture at the designated point)
- » Video display more video streams with adjustable time offset

- » Visual enhancements for each stream separately (brightness, contrast, intensity-color map)
- » Zooming (in, out, defined area)
- » Screenshot (whole picture, defined area)
- » Motion highlight (moving object is indicated, with movement vector and speed, trajectory line)
- » Motion tracking (a selected object/area can be tracked, send control information for the camera)
- » POI overlay on the map or video
- » Reporting feature (email sending with screenshot, operator comments)

### 2.8 Conclusions

The presented UAV system is under testing and is waiting for the final approval. Currently the UAV system operates with all major functions and the system parameters are continuously enhanced. The pre-serial production will be started in Q1 or 2010.

### 2.9 Acknowledgement

We thank the support of the National Office for Research and Technology.



## BXAP13 UAV features & specifications

### UAV features

- » 3D CAD design
- » Composite construction
- » Easily transportable
- » Ease of assembly and operation
- » Launch with catapult
- » Recovery without special equipment
- » Dual video stream downlink
- » Robust communication at C band
- » Onboard flight data logging

### Specifications

- » Speed: 50-140 km/h
- » Ceiling: min. 1000 m depending on configuration
- » Wing span: 3.15 m
- » Length: 1.5 m
- » Take-off weight: 7.5 kg
- » Endurance: up to 1 hour
- » Propulsion: electric brushless engine (1200W peak power)
- » Payload weight: 3 kg

### Ground Control Station

- » Whole system can be operated by the control station
- » Dual operator configuration
- » Additional operator workstations can be connected to the control station via network interface



## BXAP13 UAV features & specifications

### Payload

#### Default payload:

- » High resolution color camera head
- » Gyro stabilized pan-tilt platform
- » More than 20x optical zoom
- » Digital picture stabilization

#### Optional:

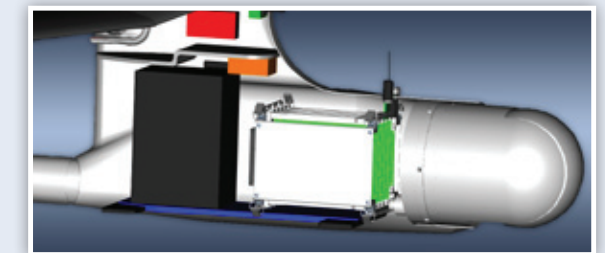
##### Thermal camera:

- » 8-12 um range
- » 384x288 pixel
- » 16 digital data
- » Up to 25 fps



#### User defined:

- » Other equipment up to 2 kg



### Onboard electronics system

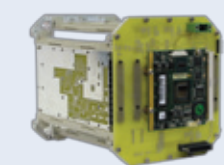
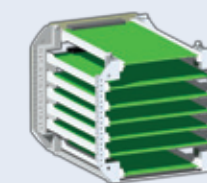
- » Rack construction
- » Autopilot
- » Communication unit
- » Video processing unit
- » Data and video logger

### Autopilot features

- » 32 bit ARM core processor
- » Flight route programming, up to 500 waypoints
- » Real time route/flight parameter modification via serial interface
- » Autonomous onboard navigation
- » Preset flight maneuvers in case of special flight events
- » 3D flight control
- » Onboard flight data logging (black box function)

### Sensors

- » Onboard Galileo-ready GPS module
- » Accurate, high speed 3D inertial navigation sensor
- » MEMS pressure sensors for airspeed and barometric altitude measurement







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